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CENTRE EUROPÉEN DE RECHERCHE ET DE FORMATION AVANCÉE EN CALCUL SCIENTIFIQUE

Multi-fidelity Optimization under Uncertainty for the Design of Complex Systems

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Introduction – Optimization under uncertainty

Optimization of complex systems under uncertainties:
Reliability-Based Design Optimization (RBDO) [1,2]



Figure – Uncertainty on the flight conditions of a sounding rocket

RANS: Reynolds Averaged Navier-Stokes



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Uncertainty sources

- > Aleatory : system environment (*e.g.*, wind gusts, atmospheric density)
- Epistemic : modelling, level of fidelity with respect to the physical phenomenon (*e.g.,* for aerodynamic simulation with "high-fidelity" RANS type code and "lowfidelity" Euler type code)



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High-fidelity (HF) codes often very expensive to evaluate: finite element codes, CFD code (e.g., RANS)



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Problem formulation $\min_{\boldsymbol{d},\boldsymbol{p}} f(\boldsymbol{d},\boldsymbol{p}) \quad s.t. \quad \begin{cases} \boldsymbol{h}(\boldsymbol{d},\boldsymbol{p}) \leq 0 \\ \mathbb{P}[\boldsymbol{g}(\boldsymbol{d},\boldsymbol{X}(\boldsymbol{p}),\boldsymbol{Z}) \leq 0] \leq P_f^T \end{cases}$

Design variables: $d \in \mathbb{R}^{n_d}$, $p \in \mathbb{R}^{n_p}$

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- Design variables: $d \in \mathbb{R}^{n_d}$, $p \in \mathbb{R}^{n_p}$
- **Random variables:** $X \sim f_{X|p}(\cdot)$ (controlled) and $Z \sim f_Z(\cdot)$ (not controlled)





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Classical RBDO techniques

Two-level approach

Optimization	Ð
Reliability Analysis	Ð

- **Reliability analysis by MCS** [3]
- Methods using a linear approximation for the reliability analysis:
 - Reliability Index Approach (RIA) [4]
 - Performance Measure Approach (PMA) [5]









Sequential Optimization and Reliability Assessment (SORA) [7]







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Useful reliability analysis techniques

Objective: estimate if $\mathbb{P}[g(X, Z) \le 0] \le P_f^T$

Use of Iso-probabilistic transformation: $\boldsymbol{u} = \mathcal{T}([\boldsymbol{X}, \boldsymbol{Z}]^T)$

- > First Order Reliability Method (FORM) [8]
 - Optimization problem to solve to find the MPFP (Most Probable failure Point):

 $\boldsymbol{u}^* = \operatorname{argmin} \|\boldsymbol{u}\| \quad s.t. \quad g(\boldsymbol{u}) = 0$

And: $\widehat{P_f} = \Phi(-\|\boldsymbol{u}^*\|)$

- > Inverse FORM (inverse problem)
 - Optimization problem to solve to find the MPTP (Minimum Performance Target Point):

 $\min_{\mathbf{u}} g(\mathbf{u}) \quad s.t. \ \left| |\mathbf{u}| \right| = \beta^{T}$

With $\beta^T = -\Phi^{-1}(P_f^T)$, and Φ^{-1} the inverse CDF of the Normal law



Figure – Illustration of FORM [9]



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Figure – Constraint function translated for the deterministic optimization [7]





Figure – Constraint function translated for the deterministic optimization [7]









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1st contribution: Bayesian Optimization

Construction of surrogate models of objective and constraint functions using Gaussian Process (GP) [10]
Optimize the sub-problems by Bayesian approach [11]
(*i.e.*, using an enrichment criterion: lower confidence bound (LCB), expected violation (EV), etc.)







Initialization (k = 0)

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Criterion adapted for RA



Initialization (k = 0) $s^{(k)} = 0$ and $Z_{MPTP} = \mu_Z$

2nd contribution: Augmented space

Construction of the surrogate model of the constraint function in the augmented (joint) space (design and random variables)

 $g(\mathbf{d}, \mathbf{X}(\mathbf{p}), \mathbf{Z})$

Reuse of information in the design of experiment (DoE) obtained in previous iterations of SORA (*e.g.*, costly model evaluations)





Figure – Illustration of the augmented space for a problem with two design variables $X_1(d_1)$ and d_2 , and a random variable *Z*. [2]



3rd contribution: Multi-fidelity

Aggregation of solvers of different level of fidelity
Construction of a multi-fidelity GP using co-kriging [12]
Adapted enrichment criterion: selection of a new point {*d^{new}*, *p^{new}*} and of the information source *l^{new}*



high-fidelity (HF) and low-fidelity (LF) code.





Sounding rocket test case

> Optimization problem

 $\begin{aligned} \min GLOW(d_0, d_1, p_0, p_1) \\ s.t. \ \mathbb{P}[Alt(d_0, d_1, X(p_0, p_1), Z_0) \leq 300 \ km] \leq 10^{-3} \end{aligned}$

GLOW: Gross Lift-Off Weight **Alt:** altitude of the sounding rocket at its trajectory apogee



Figure – Dispersion of trajectories of the sounding rocket due to uncertainties, computed with the HF and LF codes





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 $\min GLOW(d_0, d_1, p_0, p_1)$ s.t. $\mathbb{P}[Alt(d_0, d_1, X(p_0, p_1), Z_0) \le 300 \ km] \le 10^{-3}$

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Design and uncertain parameters

- *d*₀ the stage diameter
- *d*₁ the combustion chamber pressure (solid propulsion)
- p₀ the propellant mass
- p_1 the throat area
- X the uncertainty due to manufacturing
- Z_0 the uncertainty on the ascent drag coefficient



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> Solvers

- HF code: multidisciplinary code (trajectory, aerodynamics, structure, propulsion)
- LF code: HF code with simplifications on aerodynamics and structure disciplines



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Computational settings

Computational costs: $c_{LF} = 0.1$ and $c_{HF} = 1$

- Initial Design of Experiments (DoE)
 - 3 * dim points in each fidelity level (dim the dimension of each function)
- 10 repetitions of the method with differents DoEs
- Stopping criterion for the Bayesian optimizations

•
$$\|\{\boldsymbol{d}^{new}, \boldsymbol{p}^{new}\}_{(i)} - \{\boldsymbol{d}^{new}, \boldsymbol{p}^{new}\}_{(i-1)}\| \le \delta$$









Results

Table – Solution of the optimization problem

	$x^{\text{opt}} = [d_0, d_1, p_0, p_1]^{opt}$	$f_{HF}(x^{opt})$	$P_f(x^{opt})$	Coût total HF
Standard SORA	[51,1; 9,75; 438; 1,48]	683,5	10 ⁻³	4054

Double-loop approach

- > Optimizer: ~100 iterations
- > Reliability analysis by MCS: $\sim 10^5$ points to estimate P_f of the order of 10^{-3}

$ightarrow \sim 10^7$ evaluations of the HF solver are needed to solve the problem













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Conclusions and perspectives

Conclusions

- > SORA method have been investigated, which is a decoupled RBDO technique
- > 3 different contributions have been implemented:
 - > The integration of Bayesian optimization in the SORA framework
 - The construction of the surrogate models in an augmented space to reuse information through the SORA iterations
 - > The use of multi-fidelity surrogate model to combine different fidelity models
- MFB-SORA improves the efficiency of the method by reducing the computational cost, while ensuring the accuracy on the optimum of the optimization problem



Conclusions and perspectives

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Perspectives

- Compare different infill criteria for the multi-fidelity Bayesian optimization
- Test other multi-fidelity GP models (e.g., Non-linear Auto-Regressive Gaussian Process (NARGP) [14] or Linear Model of Coregionalization (LMC) [15])



Thank you for your attention!



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Appendix

Bayesian Optimization

- Construct a GP based on a DoE
- > Enrich the DoE to improve the prediction of the surrogate model
- Enrichment of the DoE:

 $x^{new} = \underset{x}{\operatorname{argmin}} LCB(x) \quad s.t. \quad EV(x) \le T$

$$\succ LCB(\mathbf{x}) = \mu_f(\mathbf{x}) - \alpha * \sigma_f(\mathbf{x})$$

$$\succ \quad EV(\mathbf{x}) = -\mu_g(\mathbf{x}) * \Phi\left(\frac{-\mu_g(\mathbf{x})}{\sigma_g(\mathbf{x})}\right) + \sigma_g(\mathbf{x}) * \phi\left(\frac{-\mu_g(\mathbf{x})}{\sigma_g(\mathbf{x})}\right)$$

- \blacktriangleright { μ_f, σ_f^2 } the posterior mean and variance of the surrogate model of f
- > $\{\mu_g, \sigma_g^2\}$ the posterior mean and variance of the surrogate model of g

Problem formulation $\min_{\mathbf{x}} f(\mathbf{x}) \quad s.t. \quad g(\mathbf{x}) \ge 0$



Figure – Illustration of the LCB enrichment criterion



Appendix

Auto-Regressive model AR1 (co-kriging)

Construction of a multi-fidelity GP, based on a hypothesis of linear dependency bewteen two successive levels of fidelity

GP HF GP LF GP corrector

- $f_t(x) = \rho_{t-1} \times f_{t-1}(x) + \gamma_t(x)$ Scalar coefficient
- Recursive construction possible if the DoEs are nested





Appendix

Multi-fidelity Bayesian Optimization

- > Construct a multi-fidelity GP based on a multi-fidelity DoE (a DoE per fidelity)
- > Enrich the DoE to improve the prediction (by selecting a new point and a fidelity level to compute it)
- Enrichment of the DoE:

$$\begin{cases} x^{new} = \underset{x}{\operatorname{argmin}} LCB_{HF}(\boldsymbol{x}) \quad s.t. \quad EV_{HF}(\boldsymbol{x}) \leq t \\ l^{new} = \arg\max_{l \in \{LF, HF\}} \frac{\sigma_{red}^2(l, \boldsymbol{x}^{new})}{cost(l)^2} \end{cases}$$

• With $cost(l)^2 = \sum_{i=1}^{l} c_i$, and with *i* the cost of each level of fidelity



Problem formulation

 $\min_{\mathbf{x}} f(\mathbf{x}) \quad s.t. \quad g(\mathbf{x}) \ge 0$

Figure – Illustration of the outputs of a high-fidelity (HF) and low-fidelity (LF) code.

